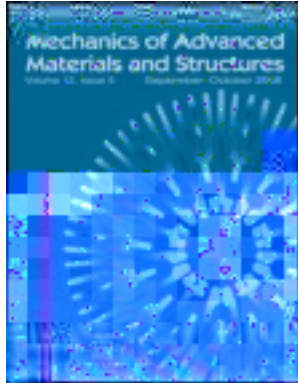


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Buckling of Functionally Graded Cylindrical Shells under Combined Loads

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By using the Ritz energy method and finite element method, buckling behaviors of combined-loaded functionally graded cylindrical shells are investigated. The combined loads are composed of axial, lateral, and torsional loads. Results show that the contribution of lateral pressure to buckling is more significant than that of axial compression or torsion and the contributions of axial compression and torsion are almost the same. Also, a practical method is proposed in this article to determine the load-dominant bound

$\frac{1}{4\pi\epsilon_0} \int_V \rho(\mathbf{r}') \frac{1}{|\mathbf{r} - \mathbf{r}'|} d\tau'$

The electrostatic energy of the charge distribution is given by

$$U = \frac{1}{2} \int_V \rho(\mathbf{r}) \phi(\mathbf{r}) d\tau$$

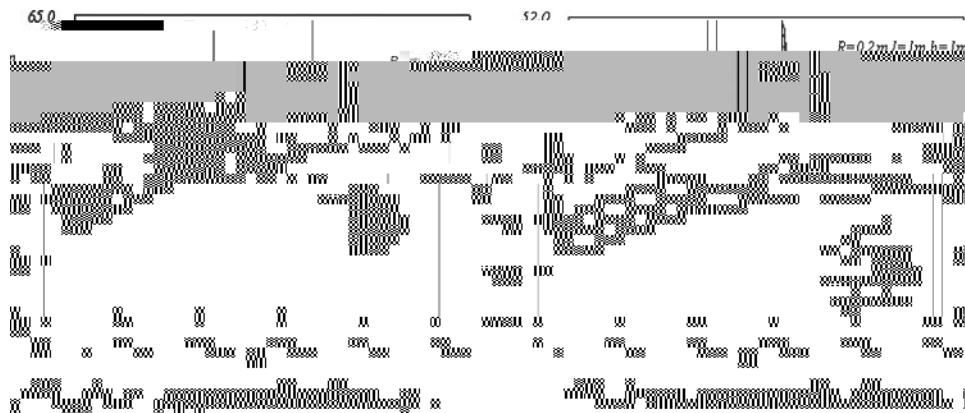
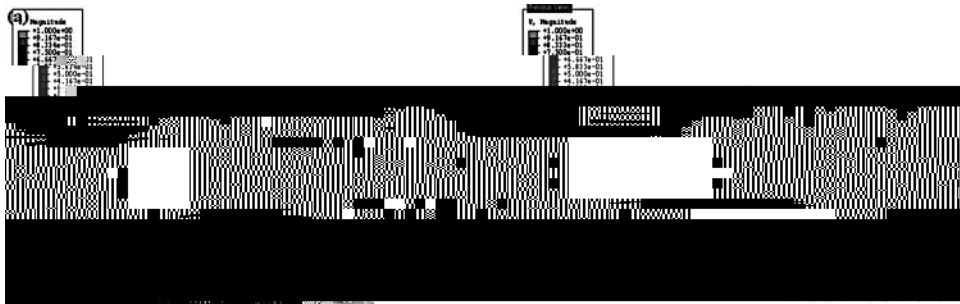


Figure 1: Comparison of the results of the finite element method with the analytical solution for the power-law exponent under

the results of the finite element method are compared with the analytical solution of the power-law exponent under the condition of $p = 1$.

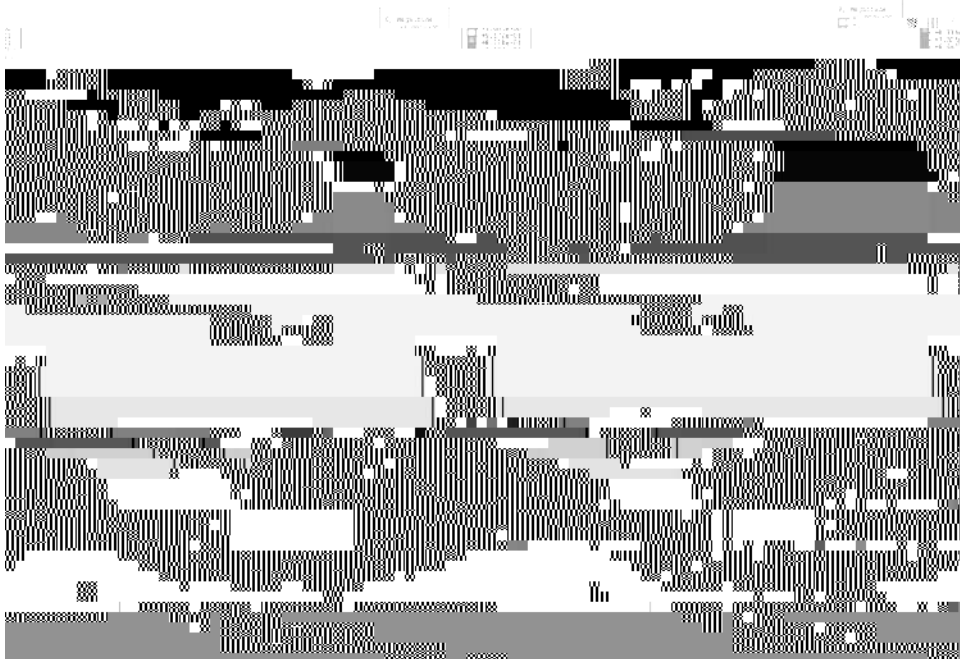
5. FEM RESULTS

The finite element method is one of the most efficient methods to solve the problem of the power-law exponent under the condition of $p = 1$. The results of the finite element method are compared with the analytical solution of the power-law exponent under the condition of $p = 1$. The results of the finite element method are compared with the analytical solution of the power-law exponent under the condition of $p = 1$. The results of the finite element method are compared with the analytical solution of the power-law exponent under the condition of $p = 1$.



(a) $A_2 = 0.01$

(b) $A_2 = 0.05$



u-k, ng modes of S surface $(N = , l = m, R = . m, h = mm)$

Figure 1 shows the magnitude of modes $u-k$ versus position x on the surface $z=0$ for $A_2 = 0.01$ and $A_2 = 0.05$. The plots show a central peak with some side structure. Figure 2 shows the magnitude of modes $u-k$ versus position x on the surface $z=0$ for $A_2 = 0.01$ and $A_2 = 0.05$. The plots show a central peak with some side structure. Figure 3 shows the magnitude of modes $u-k$ versus position x on the surface $z=0$ for $A_2 = 0.01$ and $A_2 = 0.05$. The plots show a central peak with some side structure.

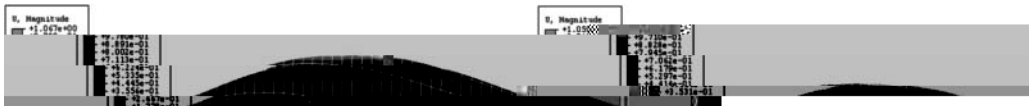


Figure 1. Magnitude of the transfer function $G(s)$ for $m=10$ (left) and $m=1000$ (right). ($N=10, l=m, R=10m, h=10mm$)

transfer function $G(s)$ is the $\frac{1}{s^2}$. If the number m under the transfer function is $m=10$, there is no limit to the frequency bounds from figure 1.

Figure 1 shows excellent agreement between theoretical results and results from the computer simulation. As the number m increases, the resonance frequency of P_{cr} shows that buckling of the structure is more sensitive to the load than to torsion. The use of $\frac{1}{s^2}$ in a second-order system, divided into three different regions, i.e., $(-\infty, -\omega_c)$, $(-\omega_c, \omega_c)$, and $(\omega_c, +\infty)$, which corresponds to resonance frequency to the system. The y-axis of buckling mode,

